

Abstract

The present invention relates to a run-in coating for gas turbines.

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The run-in coating is used for sealing a radial gap between a housing (11) of the gas turbine and rotating rotor blades (10) of same, the run-in coating (13) being applied onto the housing (11).

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According to the present invention, the run-in coating (13) is made of an intermetallic titanium-aluminum material. (Fig. 1)